TM 55-1520-228-PMD C14

CHANGE

No. 14

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D. C., 25 August 2005

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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Remove pages	insert pages
	A and B
1 and 2	1 and 2
3 and 4	3 and 4
12.1/(12.2 blank)	12.1 and 12.2
13 and 14	13 and 14

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TM 55-1520-228-PMD C13

CHANGE

No. 13

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D. C., 18 July 1997

OH-58A/C HELICOPTER
PREVENTIVE MAINTENANCE
DAILY INSPECTION CHECKLIST

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Remove pages	insert pages
17 and 18	17 and 18
18.1/(18.2 blank)	18.1/(18.2 blank)

TM 55-1520-228-PMD C 13

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TM 55-1520-228-PMD C 1 2

CHANGE

NO. 12

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 30 MARCH 1994

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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Remove pages

Insert pages

9 and 10 12.1/(12.2 blank) 15 and 16 9 and 10 12.1/(12.2 blank)

15 and 16

TM 55-1520-228-PMD C 1 2

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CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 13 April 1992

NO. 11

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

15 and 16

15 and 16

TM 55-1520-228-PMD C 1 1

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WASHINGTON, D.C., 14 February 1992

NO. 10

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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Remove pages	Insert pages
9 and 10	9 and 10
13 and 14	13 and 14

TM 55-1520-228-PMD C 1 0

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TM 55-1520-228-PMD C 9

CHANGE

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DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 19 February 1991

NO. 9

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980. is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

9 and 10 17 and 18 9 and 10 17 and 18

TM 55-1520-228-PMD C 9

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General United Army

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CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
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NO. 8

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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Remove pages	Insert pages
7 through 12	7 through 12 12.1/12.2
17 and 18	17 and 18
18.1/18.2	18.1/18.2

TM 55-1520-228-PMD C 8

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CHANGE

NO. 7

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 10 July 1989

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

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Remove pages

Insert pages

1 and 2

1 and 2

TM 55-1520-228-PMD C 7

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TM 55-1520-228-PMD C 6

CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 4 January 1988

NO. 6

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

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Remove pages

Insert pages

13 and 14

13 and 14

TM 55-1520-228-PMD C 6

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TM 55-1520-228-PMD

C 5

CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 10 November 1987

NO. 5

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

11 and 12

11 and 12

TM 55-1520-228-PMD C 5

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TM 55-1520-228-PMD

C 4

CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 23 December 1986

NO. 4

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below. New or change text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

11 through 14

11 through 14

TM 55-1520-228-PMD C 4

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TM 55-1520-228-PMD C 3

CHANGE

NO. 3

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 17 May 1985

OH-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

13 through 16

13 through 16

TM 55-1520-228-PMD C 3

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TM 55-1520-228-PMD C 2

CHANGE

No. 2

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 30 March 1984

OH-58A/C HELICOPTERS
PREVENTIVE MAINTENANCE
DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below.

Remove pages	Insert pages
7 thru 18	7 thru 18.1/18.2
19 and 20	19 and 20

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- 3. Retain this sheet in front of manual for reference purposes.

TM 55-1520-228-PMD C 2

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Major General, United States Army
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TM 55-1520-228-PMD

C 1

CHANGE

No. 1

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 8 March 1982

01-I-58A/C HELICOPTERS PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

TM 55-1520-228-PMD, 8 August 1980, is changed as follows:

1. Remove and insert pages as indicated below.

Remove pages 7 and 8 15 thru 18

Insert pages
7 and 8
15 thru 18

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- 3. Retain this sheet in front of manual for reference purposes.

TM 55-1520-228-PMD C 1

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General, United States Army

Chief of Staff

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LIST OF EFFECTIVE PAGES

Insert the latest changed pages in accordance with the instructions on the transmittal sheet.

NOTE: On a changed page, the portion of the text affected by the latest change is indicated by a vertical line in the outer margin of the page. Changes to illustrations are indicated by a letter suffix adjacent to the identification number. Added or completely revised chapters, sections, paragraphs, tables, etc., are indicated by a vertical line by the title.

Dates of issue for original and change pages are:

Original	8 August 1980	Change 8	
Change 1	8 March 1982	Change 9	19 February 1991
Change 2	30 March 1984	Change 10	14 February 1992
Change 3	17 May 1985	Change 11	
Change 4		Change 12	30 March 1994
Change 5	10 November 1987	Change 13	
Change 6	4 January 1988	Change 14	25 August 2005
Change 7	10 July 1989	-	-

TOTAL NUMBER OF PAGES IN THIS PUBLICATION IS 58, CONSISTING OF THE FOLLOWING:

Page No.	*Change No.	Page No.	*Change No.
Cover	0	9	8
A and B	14	10	12
1	0	11 and 12	8
2 and 3	14	12.1 and 12.2	14
4 through 6	0	13 and 14	14
7	8	15	
8	2	16	

^{*}Zero in this column indicates an original page.

LIST OF EFFECTIVE PAGES (Cont)

Page No.

*Change No.

17					 	 	 	 	 	 	. 9
18					 	 	 	 	 	 	13
18.	1/(1	8.2	blar	ık)	 	 	 	 	 	 	1:
19					 	 	 	 	 	 	2
20											0

^{*}Zero in this column indicates an original page.

*TM 55-1520-228-PMD

HEADQUARTERS
DEPARTMENT OF THE ARMY
Washington, D.C., 8 August 1980

OH-58A/C HELICOPTER PREVENTIVE MAINTENANCE DAILY INSPECTION CHECKLIST

GENERAL INFORMATION AND SCOPE

WARNING: CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT DAILY INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. THESE TYPE INSPECTION ITEMS ARE PRECEDED BY "MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEM."

NOTE: INDIVIDUAL INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING A DAILY INSPECTION AND MUST BE PERFORMED. THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE CRITICAL INSPECTION IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

^{*}This manual together with TM 55-1520-228-PM, 1 September 1978, supersedes TM 55-1520-228-PMS, dated 24 September 1976, including all changes.

- 1. Inspection Requirements. This manual contains complete requirements for daily inspection for OH-58 helicopters. It does not contain instructions for repair, adjustment, or other means of rectifying conditions, nor does it contain instructions for troubleshooting to find causes for malfunctioning. Specific tolerances, limits, etc., can be found in the applicable maintenance manuals. Use of the alphabetical index in the applicable manuals will facilitate locating the required information.
- **2. Maintenance Activities.** The inspections prescribed by this manual will be performed at specified periods by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot activities when required.

3. General Information.

- **a.** The inspection requirements contained herein are stated in such a manner as to establish what and when certain equipment is to be inspected and what conditions are desired/undesired. Compliance with the provisions outlined herein is required in order to assure that proper servicing has been accomplished and latent defects are discovered and corrected before malfunctioning or serious trouble results. Inspection requirements are arranged, as nearly as possible, according to the manner in which they will be performed. The requirements are divided into groups under area headings (figure 1).
- **b.** The inspection intervals designated herein will not be exceeded except in actual operational emergencies as explained here-

- in. It is the commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat, or conditions of disaster which necessitate flight to evacuate aircraft or personnel. When aircraft are operated beyond the normal inspection due-time because of such emergency situations, a circled red X status symbol and an appropriate statement (to include authority) must be entered in blocks 16 and 17 of DA Form 2408-13-1 (Aircraft Inspection and Maintenance Record) until such time as the inspection is complete. Since safety may be jeopardized when inspections are delayed to meet emergency requirements, commander will ensure that the aircraft status symbol reverts to a red "X" and that delayed inspections are accomplished immediately upon termination of the actual emergency. When unusual local conditions of environment, utilization, mission, experience of flight crew and maintenance personnel, periods of inactivity, etc., are encountered, the maintenance officer will, at his discretion, increase the scope and/or frequency of maintenance or inspections as necessary to ensure safe flight.
- **c.** This manual may contain inspection requirements applicable to specific equipment not installed on your aircraft. Those requirements should be disregarded.
- **d.** DA Form 2408-13-1 will be used to record all deficiencies or shortcomings discovered during the inspection.
- e. A 1- $\frac{1}{2}$ inch space between each area of inspection is being provided to allow insertion of additional inspection items as required by local command inspection procedures.

4. Special Instructions.

- a. A Preventive Maintenance Daily inspection is accomplished after the last flight of the day, or prior to the first flight on the next day on which the aircraft is flown. The inspection consists of visual examination and operational checks to determine that the aircraft can safely and efficiently perform the assigned mission.
- b. Work time requirements to accomplish each inspection are stated at top of the checklist.

5. Reporting Errors and Recommending Improvements.

You can help improve this manual. If you find any mistakes, or if you know of a way to improve these procedures, please let us know. Mail your letter or DA Form 2028 (Recommended

Changes to Publications and Blank Forms) located in the back of the applicable aircraft maintenance manual, (when using the 2028 from the maintenance manual, insure that the publication number number and title refer to this PMD) directly to: Commander, US Army Aviation and Missile Command, ATTN: AMSAM-MMC-MA-NP, Redstone Arsenal, AL 35898-5000. A reply will be furnished to you. You may also provide DA Form 2028 information to AMCOM via e-mail, fax, or the World Wide Web. Our fax number is: DSN788-6546 or Commercial 256-842-6546. Our e-mail address is: 2028@redstone .army.mil. Instructions for sending an electronic 2028 may be found at the back of the aircraft maintenance manual immediately preceding the hard copy 2028. For the World Wide Web use: https://amcom2028.redstone.army.mil.

6. Inspection Areas. Inspection areas are shown in figure 1.

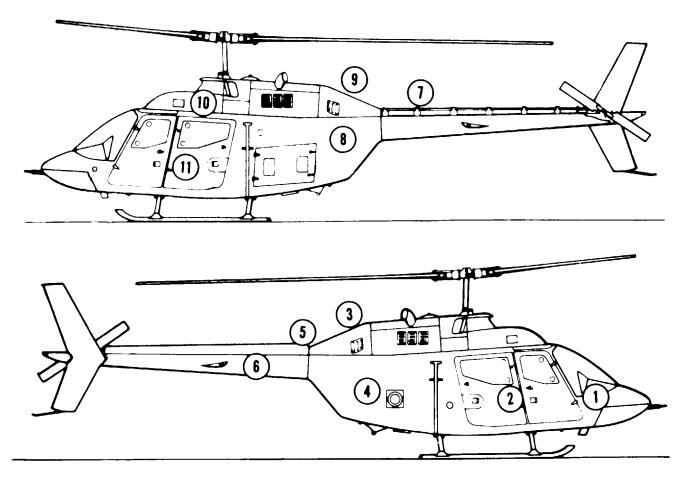


Figure 1. Inspection Areas.

Area No. 1	Nose Area	All surfaces and components in nose compartment and on exterior ahead of crew doors.
Area No. 2	Cabin Electrical Equipment Shelf and Landing Gear Area (Right Side)	All surfaces, components, and equipment inside cabin, and on cabin exterior between forward side of crew doors and aft side of passenger doors and cabin overhead. Includes complete landing gear and fuel cell sumps and filler. All the equipment in the area of the aft electrical equipment shelf,
Area No. 3	Engine Area (Right Side)	All surfaces, components, and equipment associated with engine installation, located above engine work deck.
Area No. 4	Aft Fuselage (Right Side)	All surfaces, components, below engine deck level, between cabin area and tail boom attachment bulkhead.
Area No. 5	Aft Fairing and Oil Cooler (Right Side)	All surface aft of engine cowling and oil tank.
Area No. 6	Tail Boom	All surfaces, components, and equipment located in the tail boom and vertical fin structure. Includes tail rotor, horizontal stabilizer, and control linkages. Also all sup- ports, bearings, and shafting mounted on tail boom.

Area No. 7	Aft Fairing and Oil Cooler (Left Side)	All surface aft of engine cowling, tail rotor drive shaft and oil cooler blower.
Area No. 8	Avionics and Aft Fuselage Area (Left Side)	All surfaces, components, and equipment in fuselage below engine deck level, between cabin area and tail boom attachment bulkhead.
Area No. 9	Engine Area (Left Side)	All surfaces, components, and equipment associated with engine installation, located above engine work deck.
Area No. 10	Transmission and Pylon Area	All surfaces, components, and equipment of the main rotor pylon group, from top mast to cabin roof. Includes main rotor, mast and rotating controls, transmission with accessories and mounts, servo actuators, and hydraulic system.
Area No. 11	Cabin and Landing Gear Area (Left Side)	All surfaces, components, and equipment inside cabin, and on cabin exterior between forward side of crew doors and aft side of passenger doors and cabin overhead. Includes complete landing gear.

PREVENTIVE MAINTENANCE DAILY CHECKLIST

The Preventive Maintenance Daily Checklist will be accomplished following the last flight of the day or prior to the first flight on the next day on which the aircraft is flown. The inspection consists of visual examination and operational checks to determine that the aircraft can safely and efficiently perfoom its assigned mission.

DAILY INSPECTION TOTAL WORK TIME: 2.3 Work Hours

Seq. No.	Item and Procodure	C8	Seq No.	Item and Proc.dure
	NOSE AREA No. 1		1.5	Radar warning antennas for damage and security.
1.1	Inspect aircraft forms and records for recorded discrepancies (DA PAM 738-751).		1.6	Landing lights, brackets, terminals and wiring for condition.
1.2	Nose section exterior for visible damage.		1.7	Landing/search light for damage and security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		1.8	UHF antenna for condition and security.
1.3	Pitot tube and static ports for obstructions and		1.9	Vent system drain for obstructions, condition.
	cleanliness.		1.10	Chin bubbles for damage and cleanliness.
1.4	Forward proximity warning system antenna for condition and security.		1.11	Windshields and windows for damage and cleanliness.

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and inspection.

Seq. No.	Item and Procedure	C2	Seq. No.	Item and Procedure
1.12	WSPS Windshield Deflector for damage and security.		2.8	Crew and passenger doors for positive latching and proper operation, windows and vents for
1.13	WSPS Lower Cutter Assembly for damage and security.			cleanliness and damage. Check hinges and condition of weather stripping. Door posts for
1.14	Breakaway Tip for looseness or play.			cracks.
	CABIN AND LANDING GEAR AREA No.2 (RIGHT SIDE)		2.9	Crew and passenger door jettison handles for proper installation and breakaway safetywire.
2.1	Cabin exterior for damage.		2.10	Cabin interior for cleanliness, proper stowage of
2.2	Check fuel for water and other contamination. Drain fuel sump (use sample jar) through the sump drain.		2.11	equipment and visible damage. Nose section interior for cleanliness; equipment for visible damage and loose connections.
2.3	Landing gear for condition of skid tubes, cross tubes, attachment fittings and skid shoes for evidence of hard landing damage and security.			MANDATORY SAFETY OF FLIGHT
2.4	Radar altimeter antennas for damage and security.		0.10	INSPECTION ITEM
2.5	Transponder antenna for damage and security.		2.12	Anti-torque pedals for freedom of operation and security.
2.6	Radar warning antenna for damage and security (blade).		2.13	Current compass correction card for availability and-legibility.
2.7	Marker beacon antenna for damage and security.			availability and-legibility.

C8

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
2.14	Instrument panel and instruments for security, condition and range markings.		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
2.15	FAT gage for condition and security.	2.22	Cyclic control for freedom of operation.
2.16	Check fuel valve handle-on (FWD) and off (AFT) then on (FWD) for binding, proper operation, condition, and security.	2.23	Cyclic, boots and grips for condition, security.
2.17	Overhead console for condition and security.	2.24	Copilot cyclic control for security of electrical connectors.
2.18	Ulility light and wire for condition security, and proper mounting.	2.25	Seat bottom for security, damage, and condition of webbing and date of installation. Seat back cushions for cleanliness and condition.
2.19	Heater control for condition and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	2.26	Safety belts and shoulder harness for damage, corrosion, cuts, fraying, and security. Inertia reels for damage, security, and positive locking and unlocking. Safety belt mounting hinges and brackets for cracks (visual). Check retarder springs for proper operation.
2.20	Collective and power controls for freedom of operation.	2.27	Armor panels for condition and security.
2.21	Collective boots for condition, security and throttle set screw.		

Seq No.	Item and Procedure	Seq. No.	Item and Procedure
2.28	Fire extinguisher for designated location, presence of inspection date tag, broken or missing seal, pressure indicator in green, extinguisher and brackets secure. First aid kits for designated location, presence of inspec-		ENGINE AREA No. 3 (RIGHT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
	tion date tag, broken or missing seal, completeness of side pocket contents, legible identification markings, and security (refer to TM 55-1500-204-25/1).	3.1	Engine air inlet bellmouth assembly through engine inlet fairing window for separation, cracking, deformation, security and obstructions and plenum area for loose or foreign objects.
2.30	Passenger seat and back cushions, retainer loops and tabs for damage, condition and security. Sound-proofing for condition, security, and cleanliness.	3.2	Particle separator for damage and particle ejection opening for freedom of obstructions, and loose vortex tubes (generators).
2.32	Aft electrical equipment shelf for unauthorized or foreign	3.2.1	Inspect particle separator to induction fairing for fit and sealing. Firewall to induction fairing for fit and sealing.
	objects, cleanliness, and security of components.	3.3	Engine cowling and fairing for damage, security, and condition of fasteners and hinges.
		3.4	Anti-collision light for condition and security; loose connections.
		3.5	Engine mounts and engine mount fittings, for cracks, damage and security.
		3.5.1	Remove chip detector on freewheeling assembly and check for metal particles and reinstall.
		3.6	Fuel control linkage for damage and security.

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

Seq No.	Item and Procedure.	C8	Seq No.	Item and Procedure
3.7	Compressor armor for condition and security.		3.12	Starter generator for condition. security Exhaust duct for condition, obstruction, and security
3.8	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		3.13	Main input driveshaft coupling for evidence of overheating. Exterior plating for heat discoloration or blistering. Paint strips (if used) for discoloration.
3.8	Compressor, bleed valve for damage and obstruction. MANDATORY SAFETY			MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
	OF FLIGHT INSPECTION ITEM			Inspect main rotor driveshaft 206-040-371 for damage and fastener security.
3.9	Bleed air elbows (2) for damage. Engine, accessories. and connections for damage and security.			AFT FUESELAGE AREA No. 4 (RIGHT SIDE)
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		4.1	Fuel filler cap for proper locking and condition of seal, security of lanyard.
3.11	Exhaust stacks for damage, corrosion obstructions. and security, Exhaust stack clamp for cracks, corrosion. and security.		4.1.1	If installed, inspect rubber pad below fuel receiver for debonding.

Seq No.	Item and Procedure	C8	Seq No.	Item and Procedure
4.2	FM homing antenna for condition and security.			
4.3	Battery and fuel vents for condition, obstruction, proper positioning and security.			
4.4	Air ducts for condition and security.			
4.5	External power access door for condition and security.			
4.6	Radar warning antennas for damage and security.			

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Seq No.	Item and Procedure	Seq No.	Item and Procedure
5.1	AFT FAIRING AND OIL COOLER AREA No. 5 (RIGHT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Engine oil tank for damage, security and servicing	6.3	Long tail rotor drive shaft and bearings for evidence of overheat; inspect slippage marks, excessive grease leakage, bonded fittings for security, splined adapters for freedom of movement. Do not wipe grease from seal area. Inspection not required after compliance with MWO 55-1520-228-50-25.
3.1	lines for leaks and damage. Loose or missing identification tape on oil lines.	6.3.1	Check segmented driveshaft for condition, shaft coupling disc for distortion, cracks and security, splined
5.2	Oil filler access door for condition and security.		adapter for freedom of movement, and bearing and driveshaft for evidence of over heating. Do not wipe
5.3	Engine external scavenge oil filter for damage, security, lines for leaks and damage, and bypass indicator (red		grease from seal area. Required after compliance with MWO 55-1520-228-50-25.
	button) in. Loose or missing identification tape on oil lines. After compliance with MWO 55-1520-228-50-44.	6.4	Horizontal stabilizer for visible damage.
	TAIL BOOM AREA No. 6	6.5	Navigation lights for condition and security.
	MANDATORY SAFETY OF FLIGHT	6.6	VOR antennas for damage and security.
	INSPECTION ITEM	6.7	Tail light and support for damage and security.
6.1	Tail boom exterior for visible damage.	6.8	Vertical stabilizer, tail skid, and antenna leads for security
6.2	Driveshaft cover for damage, security, and condition of fasteners.		and damage.

C14

Seq No.	Item and Procedure	Seq No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		
6.8.1	Inspect the inboard skin of the vertical fin for cracks originating from the four attachment inserts. None allowed.		
6.9	Ballast installation for security and condition.		
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		
6.10	Tail rotor controls for damage and security of attachment bolts .		

"FOD REMINDER"

Seq No.	Item and Procedure	Seq No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		in the area approximately 7 inches outboard from the butt end and 1.5 inches aft of the leading edge. If inspec-
6.10.1	Tail rotor pitch links for cracks using ten-power magnifi- cation. Pay particular attention to area around ball bear- ing at each end of pitch link. Suspected cracks shall be confirmed by penetrant inspection.		tion reveals a crack, blade must be removed for eval- uation prior to next flight. Tail rotor blades with serial number TLL-8000 and below require ten-power magnifi- cation for the inspection and careful inspection for corro- sion and deterioration of clear coating on areas left clear
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		for inspection. Not required after compliance with MWO 55-1520-228-50-25.
6.11	Tail rotor gear box for security, oil level, condition, and vent for obstruction. Check light glass for cracks, damage, and stains that might give a false indication of the oil level.	6.13.1	Inspect both tail rotor blades for cracks using a ten-power magnification, the skin surface area where the root-end trailing edge balance weight is riveted (both sides). If inspection reveals a crack, blade must be removed prior
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	0.44	to next flight.
6.12	Tail rotor hub and blade assembly for security and dam-	6.14	Vulnerability reduction control system access panel for damage and security.
	age.	6.15	VOR antenna for damage and security.
6.12.1	Tail rotor blades for cleanliness and clean as required.	6.16	Horizontal stabilizer for visible damage
6.12.2	Inspect all self-locking nuts on the tail rotor blade and hub assembly for cracks, damage and security.	6.17	Navigation lights for condition and security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
6.13	Inspect both tail rotor blades for cracks on both sides	6.18	Inspect tail rotor blade tip blocks for cracks, debonding, loose, missing or corroded rivets.

Seq No.	Item and Procedure	Seq No.	Item and Procedure
	AFT FAIRING AND	8.2	ADF sense antenna for damage and security.
	OIL COOLER AREA No. 7 (LEFT SIDE)	8.3	ADF loop antenna for damage and security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	8.4	Avionics compartment door for condition, security, proper operation.
7.1	Engine oil tank for damage, security and servicing lines for leaks and damage.	8.5	Avionics compartment for loose or unauthorized equipment.
7.2	Oil cooler blower and drive shaft for damage, cracks, and security. Blower air intake for foreign material and clogging.		(a) Inspect battery box installation aft of station 120 for distortion or cracks in fuselage frames and adjacent structure. After compliance with MWO 55-1520-228-50-27 or MWO 55-1520-228-50-37.
7.3	Aft fairing assembly for damage and condition of fasteners. AVIONICS AND AFT No. 8		(b) Inspect rack installation at and behind station 167 for distortion or cracks in fuselage frames and adjacent structure. After compliance with MWO 55-1520-228-505-27 or MWO 55-1520-228-50-37.
	FUSELAGE AREA	8.6	Foam pad for condition, cleanliness, and security.
	(LEFT SIDE)	8.7	Battery for security and leakage.
8.1	Aft proximity warning system antenna for damage and security.	8.8	External power receptacle for condition and security.
	_	8.9	FM homing antennas for damage and security.

C12

Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
	ENGINE AREA No, 9 (LEFT SIDE)	9.8	Linear actuator, linkage, and electrical lead for condition and security.
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	9.9	Engine automatic relight for condition and security.
9.1	Engine air inlet bellmouth assembly through engine inlet fairing window for separation, cracking, deformation, security and obstructions and plenum area for loose or	9.10	Exhaust stacks for damage, corrosion, obstructions, and security. Exhaust stack clamp for cracks, corrosion, and security.
	foreign objects.	9.11	Engine combustion chamber housing, turbine support assembly and exhaust ducts for cracks, dents, burned, or
9.2	Particle separator for damage and particle ejection opening for freedom of obstructions.		buckled areas.
9.3	Engine cowling and fairing for damage, security, and condition of fasteners and hinges.	9.12	Drain valve for leakage, security and contact with drive shaft.
9.4	Engine mounts and engine mount fittings, for cracks, damage and security,		MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
9.5	Compressor armor for condition and security.	9.13	Fuel nozzle and lines for security and leaks.
9.6	Airframe mounted fuel filter for condition, leaks and security. After compliance with MWO 1-1520-228-50-48.	9.14	Forward tail rotor driveshaft for cleanliness, corrosion, and freedom of movement on the splined adapters.
9.7	Engine, accessories, and connections for damage and security.		

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Seq. No.	Item and Procedure	Seq. No.	Item and Procedure
10.1	TRANSMISSION AND PYLON AREA No. 10 (LEFT SIDE) MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Main rotor hub assembly for leakage, damage, securi-		surfaces. If inspection by flashlight reveals a crack indication, validate the crack indication using fluorescent penetrant. If crack is confirmed, remove yoke assembly from service. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
	ty, and reservoirs for proper servicing.	10.6	Main rotor blades for condition, damage, and security.
10.2	Sight glass for damage and stains which might give false oil level indication. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Check pillow blocks for mounting condition, security of mounting bolts and slippage marks. If slippage has occurred, replace hardware. Static stops for damage and security. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM	10.6.1 10.7 10.8	Main rotor blades for cleanliness and clean as required. Main rotor blades for corrosion, specifically in the area of the inertia weight retention screw holes. Tip cap firing for condition and security. Check main rolor blades for protective coating, clean blades and re-wax as required. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Visible areas of mast for damage.
10.5	Main rotor yoke assembly for cracks (visual) between the pillow block bores and lower flange on inboard surfaces. Wipe clean yoke web section inner and outer		

Boot for damage and security.			
, , , , , , , , , , , , , , , , , , ,			MANDATORY SAFETY OF FLIGHT INSPECTION ITEM
MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Swashplate, lever and sleeve, and connecting linkage		10.12.1	Main rotor pitch links for damage and security of attachment, tubes for corrosion and cracks; pay particular attention to swagged ends at jam nut.
for security and visible damage. Swashplate support base drain holes open and free of foreign matter.		10.13	Hydraulic servo actuator support assembly for cracks, corrosion, security and damage.
NOTE Ensure swashplate is level for inspection of outer- ring self-aligning bearings for main rotor push-pull		10.14	Hydraulic system components and lines for security, chaf- ing, damage, leaks and reservoir for servicing. Collective actuator bellcrank cover for missing rubber flipper or flip- per bond failure. Proper security of hydraulic reservoir cap
MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10.15	and pin assembly. (Refer to TM 55-1520-228-23-1). Hydraulic filter element indicator for filter clogged indication.
Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT REMOVE COTTER PIN).		10.16	Edges of main drive shaft cover assembly for evidence of grease leakage. Security of attaching hardware.
fe k	Swashplate, lever and sleeve, and connecting linkage or security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outering self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT	Swashplate, lever and sleeve, and connecting linkage or security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT)	Swashplate, lever and sleeve, and connecting linkage or security and visible damage. Swashplate support base drain holes open and free of foreign matter. NOTE Ensure swashplate is level for inspection of outerring self-aligning bearings for main rotor push-pull tubes. MANDATORY SAFETY OF FLIGHT INSPECTION ITEM Cyclic and collective control linkage for security and condition. Check for security of attachment bolts by attempting to torque nuts with fingers (DO NOT) 10.13

Seq. No.	Item and Procedure	C13	Seq. No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10.22	Pylon isolation mount cover for condition, drag pin assembly to static stop or cabin roof for foreign objects and damage.
10.17	Power turbine governor (N2) linkage for freedom of operation, security, and condition.		10.22.1	Wipe up oil spillage and clean area using cloth dampened with dry cleaning solvent. Wipe dry with clean cloth. Visually examine clevis area and mount
10.18	Sight glass for damage or stains which might give false oil level indication.			for cracks, nicks, scratches, and loose or missing screws, bolts, and cotter pins.
10.19	Rotor tachometer generator, hydraulic pump, for condition and security.		10.23	Inspect area under drag pin bearing for oil and water.
10.20	Transmission oil cooler and duct for condition, security and obstruction.		10.24	Inspect transmission deck area for dents, cracks, holes, and overall
10.21	Transmission oil filter and filter head for condition, security, and leaks.			condition.
10.21.1	Roof mounted transmission oil filter for damage, security, lines for leaks and		10.25	Pylon support links for cracks, condition (visual) and security of hardware.
	damage, and bypass indicator (red button) in. After compliance with MWO 1-1520-228-50-51.		10.26	Inspect Electrical wiring for damage, security, and corrosion.

Seq. No.	Item and Procedure	C 13	Seq. No.	Item and Procedure
	MANDATORY SAFETY OF FLIGHT INSPECTION ITEM		10-33	FM No. 2 antenna for damage and security.
10.27	Transmission and connections for damage and security. Transmission for		10-34	WSPS Upper Cutter Assembly for damage and security.
	servicing. (Filler cap for security.)		10-35	GPS antenna mounted on WSPS upper cutter assembly for damage and
10.28	Particle separator for damage and particle swirls for condition and security.			security. After compliance with MWO 1-1520-228-50-53.
10.29	Check security of engine cleaning provision tube and cap assembly.			
10.30	Transmission faking for damage and condition of fasteners.			
10.31	IFF antenna for damage and security.			
10.32	Glide slope antenna for damage and security.			

Seq. No.	Item and Procedure	C2	Seq. No.	Item and Procedure
	CABIN AND LANDING GEAR AREA No. 11			POWER ON
	(LEFT SIDE)		1.15	Pitot heater for operation.
11.1	Landing gear for condition of skid tubes, cross tubes, attachment fittings and skid shoes for evidence of hard landing damage and security.		1.16	Warning and caution panel lights for illumination on test.
11.2	Crew and passenger doors for positive latching		2.33	Interior and instrument panel lights for proper operation.
11.2	and proper operation, windows and vents for cleanliness and damage. Check hinges and condition of weather stripping.			WARNING
11.3	Seats for security, damage, and condition of webbing. Cushions for cleanliness and condition and installation date.			Assure defogging blower motors are free of foreign debris prior to energizing motors.
11.4	Armor plating for condition and security.		2.34	Operate the defogging blower motors for a minimum of 15 seconds to insure proper func-
11.5	Crew and passenger door jettison handles for proper installation and breakaway safety wire.		All Areas	tioning. Exterior lights for proper operation.
	LUBRICATION			
All Areas	Service in accordance with TM 55-1520-228-23, applicable to the daily requirements.			

By Order of the Secretary of the Army:

E. C. MEYER

General, United States Army

Chief of Staff

Official:

J. C. PENNINGTON

Major General, United States Army
The Adjutant General

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